



WorldView 1

WorldView-1 provides earth imagery at 50cm spatial resolution. Being a panchromatic(Black & White) sensor, WorldView-1 can acquire very large areas in short period of time. This sensor was launched by DigitalGlobe on September 18, 2007. With an average revisit time of 1.7 days, WorldView-1 is capable of collecting up to 750,000 square kilometers (290,000 sq. mi) per day of half-meter imagery. WorldView-1 satellite sensor was successfully launched from Vandenberg Air Force Base, California, U.S.A.



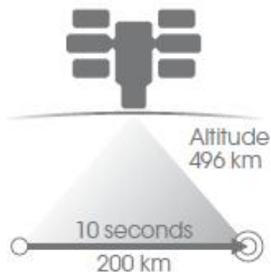
Below are the specifications of WorldView-1 sensor:

Launch Information:	Date: September 18, 2007 Launch Vehicle: Delta II 7920 Launch Site: Vandenberg Air Force Base
Orbit:	Altitude: 496 kilometers Type: Sun synchronous, 10:30 am descending node Period: 94.6 minutes
Sensor Bands:	Panchromatic
Sensor Resolution (GSD Ground Sample Distance):	0.50 meters GSD at nadir 0.59 meters GSD at 25° off-nadir
NIIRS Equivalency:	NIIRS potential of greater than 5.0
Dynamic Range:	11-bits per pixel
Swath Width:	17.6 kilometers at nadir
Pointing Accuracy & Knowledge:	Accuracy: <500 meters at image start and stop
Knowledge:	Supports geolocation accuracy below
Retargeting Agility:	Acceleration: 2.5 deg/s/s Rate: 4.5 deg/s Time to slew 300 kilometers: 10.5 seconds
On-board Storage:	2199 gigabits solid state with EDAC
Max Viewing Angle / Accessible Ground Swath:	Nominally +/-45° off-nadir = 1036 km wide swath Higher angles selectively available
Per Orbit Collection:	331 gigabits
Max Contiguous Area Collected in a Single Pass:	60 x 110 km mono 30 x 110 km stereo
Revisit Frequency:	1.7 days at 1 meter GSD 4.6 days at 25° off-nadir or less (0.59 meter GSD)
Geolocation Accuracy (CE90):	Geolocation Accuracy specification of 6.5m CE90% at nadir, with actual accuracy in the range of 4.0 – 5.5m CE90%, excluding terrain and off-nadir effects

Ball Aerospace provided its BCP 5000 spacecraft bus, and the WorldView 60 (Similar to the one used by QuickBird sensor) camera for the commercial imaging satellite. A new feature of the WorldView-1 spacecraft are CMG (Control Moment Gyroscopes) actuators for precise and highly responsive pointing control. The BCP-5000 bus provides increased power, stability, agility, data storage and transmission (over the BCP-2000 bus) as the demand for Earth remote-sensing information becomes more comprehensive.

The S/C is 3-axis stabilized. The ADCS (Attitude Determination and Control Subsystem) employs star trackers, IRU (Inertial Reference Unit) and GPS for attitude sensing, and CMGs as actuators. A S/C body-pointing range of $\pm 40^\circ$ about nadir is provided corresponding to a FOR (Field of Regard) of 775 km in cross-track. An instantaneous pointing accuracy of ≤ 500 m is provided at any start and stop of an imaging sequence. On the ground, the geolocation accuracy of the imagery is 5.8 to 7.6 m without GPCs (Ground Control Points) and 2 m with GPCs (3σ). The agile S/C provides retargeting at a rate of 4.5%/s with an acceleration of 2.5 $^\circ$ /s²; it takes 9 s to slew the S/C over a ground distance of 300 km.

Altitude and Slew Time



Collection Scenarios

(30° off-nadir angle)

